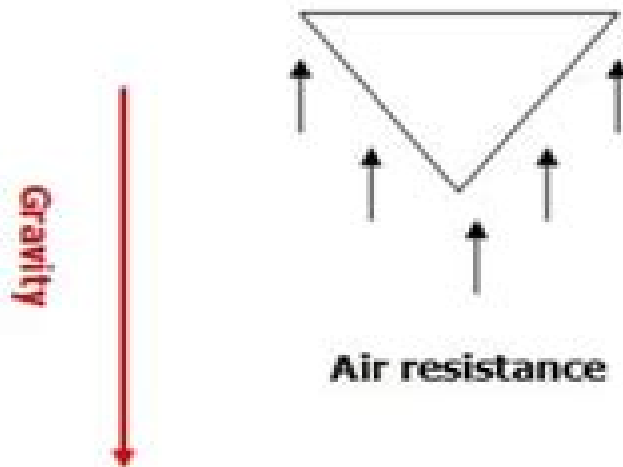


Diagram 1.

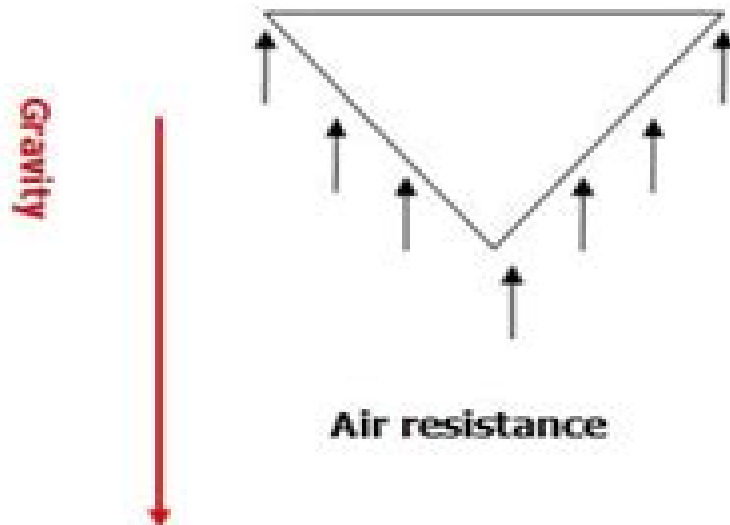
Parachute with small surface area



Less particles hit so
the parachute fell
faster.

Diagram 2.

Parachute with large surface area



More particles hit so
the parachute fell
slower.

Gcse Physics Parachute Experiment Mass

KJ Lindholm-Leary



Gcse Physics Parachute Experiment Mass:

AQA GCSE (9-1) Physics Student Book Nick England, Steve Witney, 2016-08-01 Exam Board AQA Level GCSE Subject Physics First Teaching September 2016 First Exam June 2018 AQA approved Apply and develop your students knowledge and understanding of Physics with this textbook that builds mathematical skills provides practical assessment guidance and supports all the required practicals Provides support for all the required practicals with activities that introduce practical work and other experimental investigations in Physics Builds understanding and knowledge with a variety of questions to engage and challenge Test Yourself questions Show You Can challenges Chapter review questions and synoptic practice questions Supports Foundation and Higher tier students in one book with Higher tier only content clearly marked Builds Literacy skills for the new specification with key words highlighted and practice extended answer writing and spelling vocabulary tests **FREE GCSE SCIENCE TEACHER GUIDES** These will be provided for free via our website To request your free copies please email science@hodder.co.uk **GCSE Success AQA Additional Science Revision Guide** HarperCollins Publishers Limited, 2006-09 Helps students manage their revision and prepare for exams efficiently This title offers content that is broken into manageable sections It provides exam tips and techniques to support students in the revision process

My Revision Notes: WJEC GCSE Science Double Award Adrian Schmit, Jeremy Pollard, 2018-01-15 Exam Board WJEC Level GCSE Subject Science First Teaching September 2016 First Exam Summer 2018 Target success in Science with this proven formula for effective structured revision key content coverage is combined with exam style tasks and practical tips to create a revision guide that students can rely on to review strengthen and test their knowledge With My Revision Notes every student can Plan and manage a successful revision programme using the topic by topic planner Consolidate subject knowledge by working through clear and focused content coverage Test understanding and identify areas for improvement with regular Now Test Yourself tasks and answers Improve exam technique through practice questions expert tips and examples of typical mistakes to avoid Get exam ready with extra quick quizzes and answers to the practice questions available online Please note that some of the quizzes from the WJEC GCSE My Revision Notes series are also used in the WJEC GCSE Teaching and Learning resources **My Revision Notes: Edexcel International GCSE (9-1) Physics** Ian Horseywell, 2018-11-05 Target success in Edexcel International GCSE Physics with this proven formula for effective structured revision key content coverage is combined with exam style tasks and practical tips to create a revision guide that students can rely on to review strengthen and test their knowledge Plan and manage a successful revision programme using the topic by topic planner Consolidate subject knowledge by working through clear and focused content coverage Test understanding and identify areas for improvement with regular Now Test Yourself tasks and answers Improve exam technique through practice questions expert tips and examples of typical mistakes to avoid Get exam ready with extra quick quizzes and answers to the practice questions available online **GCSE Physics for You** Keith Johnson, 1986 **My Revision Notes:**

CCEA GCSE Physics Roy White, 2018-01-15 arget success in CCEA GCSE Chemistry with this proven formula for effective structured revision key content coverage is combined with exam style tasks and practical tips to create a revision guide that students can rely on to review strengthen and test their knowledge With My Revision Notes every student can Plan and manage a successful revision programme using the topic by topic planner Consolidate subject knowledge by working through clear and focused content coverage Test understanding and identify areas for improvement with regular Now Test Yourself tasks and answers Improve exam technique through practice questions expert tips and examples of typical mistakes to avoid Answers to the practice questions available online

Experimental Determination of Parachute Apparent Mass and Its Significance in Predicting Dynamic Stability T. Yavuz, D. J. Cockrell, American Institute of Aeronautics and Astronautics, 1981 Experimental Investigation of Added Mass During Parachute Deceleration Brad Charles Cochran, 1991

Theory and Experiment on Parachute Opening Shock and Filling Time Helmut G. Heinrich, University of Minnesota. Department of Aerospace Engineering and Mechanics, Royal Aeronautical Society, 1971 A method of parachute opening force and filling time calculation is presented which is based on the time functions of aerodynamic drag included air mass apparent mass and effective porosity Using the momentum and continuity equations an idealized canopy shape development and a linear drag area time relationship expressions are obtained for the canopy volume and velocity The calculated peak forces agree well with at least mean square averages of field test results of 28 ft solid flat parachutes deployed at speeds from 175 to 450 fps at altitudes of 6 000 13 000 and 21 000 ft and with suspended weights between 200 and 820 lb **Experimental Determination of Parameters for the Calculation of Parachute Filling Times** Rudi J. Berndt, Air Force Flight Dynamics Laboratory (U.S.), Wissenschaftliche Gesellschaft für Luft- und Raumfahrt, Deutsche Gesellschaft für Raketentechnik und Raumfahrt, 1964 The filling or inflation process of a flat circular solid cloth type parachute canopy operating under finite mass conditions is analyzed in detail Experimental test results acquired during model tests in wind tunnels and during full scale tests on personnel and air drop type parachute canopies at altitudes between sea level and 21 000 feet form the basis for the analysis Parameters and relationships acquired as a result of the experimental test programs or derived analytically are expressed in mathematical terms so that they may be used directly for the solution of the mass balance and instantaneous velocity equations upon which the calculations of parachute filling time is based The parameters and relationships concern the shape of the inflating canopy the growth of the drag area of the canopy the canopy flow velocities and growth of the air volume enclosed by the drag producing surface and the instantaneous velocity of the load parachute configuration **Flight Test of a 31.2-foot-diameter Modified Ringsail Parachute Deployed at a Mach Number of 1.39 and a Dynamic Pressure of 11.0 Pounds Per Square Foot** John S. Preisser, Clinton V. Eckstrom, Harold N. Murrow, 1967 **An Experimental Investigation of Model Parachutes Under Finite Mass Conditions** Herbert R. Brown, AIR FORCE INST OF TECH WRIGHT-PATTERSON AFB OHIO SCHOOL OF ENGINEERING., 1975 A two part

experimental investigation was undertaken The first part was to develop testing techniques for a finite mass test fixture The test fixture is designed such that a model parachute and load are free to move along a wire track through a wind tunnel test section Also a servomechanism simulated the component of force due to gravity acting on the load The second part was to test the feasibility of using mass ratios as similarity parameters for wind tunnel testing of model parachutes Tests were conducted to determine the friction force of the wire track on the test load of varying weights and for varying degrees of lubrication Graphite lubrication of the wire track reduced the friction force and adjustments to the servomechanism were developed which compensate for this reduced friction force

Experimental Studies for Determining the Inflation Characteristics of a Parachute Under Finite Mass Conditions Robin Edward Schaller, 1963

Experimental

Determination of Transonic and Supersonic Pressure Distribution and Mass Flow Accommodation for the Hemisflo Ribbon Parachute During Inflation Ronald J. Niccum, Air Force Flight Dynamics Laboratory (U.S.), University of Minnesota. Department of Aeronautics and Engineering Mechanics, 1965

This report presents the results of an experimental study to qualitatively determine the air flow through a Hemisflo ribbon parachute during the inflation process These measured mass flows are analyzed relative to free stream velocity and density ratio The results are presented as ratios of measured to inviscid flow through the canopy inlet area thus giving a measure of the canopy's ability to accommodate the air mass entering the parachute mouth Related to the mass flow studies was the determination of pressure distributions on the internal and external canopy surfaces Measurements were made at Mach 0.8, 1.2 and 3.0 In addition several tests were conducted using an ogive forebody at two upstream locations

Analysis of an Inflating Subsonic Reefed Parachute with Experimental Mass Flow Study William K. Lockman, 1963

Experimental Determination of Transonic and Supersonic

Pressure Distribution and Mass Flow Accommodation for Ribbon Parachutes During Inflation. Vol. I--the 27% Geometric Porosity Hemisflo Ribbon Parachute Ronald J. Niccum, Keith J. Goar, Warren W. Lenius, MINNESOTA UNIV

MINNEAPOLIS DEPT OF AERONAUTICS AND ENGINEERING MECHANICS., 1965 The measured mass flows are analyzed relative to free stream velocity and density ratio The results are presented as ratios of measured to inviscid flow through the canopy inlet area thus giving a measure of the canopy's ability to accommodate the air mass entering the parachute mouth Related to the mass flow studies was the determination of pressure distributions on the internal and external canopy surfaces Measurements were made at Mach 0.8, 1.2 and 3.0 In addition several tests were conducted using an ogive forebody at two upstream locations The wake of the ogive forebody with respect to pressure is analyzed Author

Program Plan for Measurement of Apollo Parachute Pressure, Strain, Load Shape, and Added Mass M. R. Bottoroff, Northrop Corporation.

Ventura Division, Manned Spacecraft Center (U.S.), 1969 The results of a study on parachute measurements conducted as part of an overall study on Apollo parachute analysis methods are presented The experimental measurements required to implement improved load and stress prediction methods are reviewed Information on several currently available pressure

transducers is presented and a number of conceptual techniques for measuring load and strain in canopy structural members are discussed. The added mass concept in parachute dynamic analysis is discussed and wind tunnel and flight test techniques for measuring added mass are indicated. A test plan for Apollo parachutes is presented consisting of a laboratory phase, a wind tunnel phase and a flight test phase.

Dynamics of Parachute Inflation H. G. Heinrich, 1969. The study is concerned with identification of significant terms in the process of parachute inflation. For finite mass cases the equations of motion for parachutes inflating in free air and in wind tunnel experiments are established and organized in view of the nonsteady terms of canopy size, systems velocity, and apparent masses and the time derivatives of these terms. Wind tunnel experiments are described, the results of which yield graphical and numerical time functions for the significant terms of the equation of motion and the force time histories. The wind tunnel results are compared with full size test information and certain identities are shown. The significant terms are combined to functions unique for the test conditions and substituted in the equation of motion. Force time functions so derived compare satisfactorily with measured force histories and the contributions of the individual terms to the instantaneous parachute force can be seen.

Determination of the Mass Flow Through Parachute Canopies in Subsonic and Supersonic Flow R. J. Niccum, The success of any parachute system is measured by its performance during the opening sequence and its performance during steady state operation. Earlier presentations in this course have shown that the parachute opening time and opening force are important dynamic parameters which may be used in determining a parachute's desirability for certain applications. Similarly the steady state aerodynamics such as tangent force coefficient, normal force coefficient and moment derivative with respect to angle of attack are among the parameters employed for gauging a parachute's steady state worthiness. The purpose of this presentation is to bring to your attention still another parameter that is useful in the design and evaluation of the performance characteristics of parachute systems, namely the mass flow through a canopy both during its inflation and its steady descent.

Note on Analysis of the Opening Shock of Parachutes at Various Altitudes Theodore Von Kármán, United States Army Air Forces Scientific Advisory Group, High decelerations up to 33g were measured on parachutes launched at 40 000 ft altitude. This appears surprising since the theoretical prediction was made that the deceleration shock of a parachute launched at a true airplane speed proportional to the terminal velocity of a falling man should be independent of the altitude. It is shown that this theory is in error because it does not take into account the so-called apparent mass of the parachute, i.e. the inertia of the air moving with the parachute. This additional mass is of course proportional to the density of the air at the given altitude and many times larger than the mass of the parachute itself, even considerably larger than the mass of the man. In this note the attempt is made to give the correct methods of analysis. Certain proposals are made for further experimentation.

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